

NATIONAL EXAMINATIONS

December 2016

07-MEC-B3 ENERGY CONVERSION AND POWER GENERATION

Three hours duration

Notes to Candidates

1. This is a **Closed Book** examination.
2. Examination paper consists of two Sections. **Section A is Calculative** with four (4) questions and **Section B is Descriptive** with two (2) questions.
3. Note that Question 4 is on two pages.
4. **Do three (3) questions (including all parts of each question) from Section A (Calculative) and one (1) question from Section B (Descriptive).**
5. **Four questions constitute a complete paper.** (Total 60 marks).
6. **All questions are of equal value.** (Each 15 marks).
7. If doubt exists as to the interpretation of any question or in the event of missing data, the candidate is urged to submit, with the answer paper, a clear statement of any assumptions made.
8. If any initial parts of a multi-part question cannot be solved the remaining parts may be worked by making appropriate assumptions for the first parts from the technical data given.
9. **Read the entire question before commencing the calculations** and take note of any hints or recommendations given.
10. Candidates may use one of the approved **Casio** or **Sharp** calculators.
11. **Reference data** for particular questions are given on pages 9 to 14. **All pages used are to be returned with the answer booklet showing where data has been obtained.**
12. **Reference formulae and constants** are given on pages 15 to 18.
13. **Steam Tables** from "Thermodynamics and Heat Power" are provided.

SECTION A CALCULATIVE QUESTIONS

Show all steps in the calculations and state the units for all intermediate and final answers.

QUESTION 1 REGENERATIVE GAS TURBINE

The initial part of this question is to determine the efficiency of a simple cycle gas turbine. The latter part is to determine the efficiency of a regenerative cycle gas turbine operating within the same boundary conditions. These conditions are as follows:

Compressor inlet air pressure	0.1 MPa	(atmospheric)
Compressor inlet air temperature	15°C	(ambient)
Turbine inlet gas pressure	0.6 MPa	(pressure ratio = 6)
Turbine inlet gas temperature	760°C	(maximum permitted)
Air mass flow rate	1 Mg/s	
Compressor efficiency	86%	
Turbine efficiency	89%	
Regenerator effectiveness	75%	
Fuel heating value	40 MJ/kg	

Assume that the gas mass flow rate is the same as that of the air (neglect fuel mass flow rate). Assume also cold air standard conditions ($k = 1.4$).

- Sketch the simple cycle on a T-s diagram and label all key points in the cycle that will be used to identify calculated values. Allow space to show regenerative cycle conditions. (1)
- Calculate the temperatures (ideal and actual) at the compressor and turbine inlets and outlets. (6)
- Calculate the net power produced by the machine and the efficiency of the simple cycle. (3)
- Consider the addition of a regenerative heat exchanger to the simple cycle with the effectiveness (actual heat transfer / maximum possible heat transfer) as given. Show the transfer of heat on the T-s diagram in (a) above.
- Calculate the rate of heat transfer in the regenerative heat exchanger. (1)
- Calculate the cycle efficiency with the regenerative heat exchanger and comment on its usefulness as an addition to the simple cycle. (2)
- Calculate the air/fuel ratio for both the simple cycle and the regenerative cycle. (2)

[15 marks]

QUESTION 2 COMBINED CYCLE PLANT

This question follows from Question 1 but can be completed without having solved any parts of Question 1.

The question considers the addition of a steam cycle (bottoming cycle) to the simple cycle gas turbine in Question 1. Exhaust gas from the gas turbine is used in a heat recovery boiler to generate steam for a steam turbine. Cycle conditions are as follows:

Gas turbine exhaust temperature	392°C	(regenerator inlet)
Regenerator outlet temperature	177°C	
Gas mass flow rate	1 Mg/s	
Fuel mass flow rate	13 kg/s	(gas turbine)
Fuel heating value	40 MJ/kg	
Gas cycle efficiency	28%	(simple cycle)
Steam pressure	1.6 MPa	
Feedwater inlet temperature	40°C	
Steam outlet temperature	350°C	
Steam turbine efficiency	88%	

Assume cold air standard conditions for exhaust gas. Assume also no heat loss in the heat recovery boiler. Use steam tables for steam conditions.

- Sketch the temperature profiles of the gas and steam on a temperature versus path length diagram and label all key points in the cycle that will be used to identify calculated values. Sketch also the turbine expansion line on an h-s diagram and label all key points to identify calculated values (2)
- Determine all enthalpies at key points in the steam cycle. (4)
- Calculate gas turbine power output and steam turbine power output and hence the combined cycle efficiency. (6)
- Calculate the temperature difference between the exhaust gas and the steam circuit at the pinch point. (1)
- Explain the significance of the pinch point with respect to the design of the heat recovery boiler. Consider primarily heat transfer and capital cost.

OR

Explain how the steam cycle could be modified to improve the thermodynamic performance of the heat recovery boiler and hence efficiency of the cycle. (2)

[15 marks]

QUESTION 3 INTERNAL COMBUSTION ENGINE

Refer to the Examination Paper Attachments **Volkswagen Jetta 2003 Model** on Page 9.

Following the steps given below, estimate the power output of the 2.0 L Four Cylinder In-line Engine as used on the Volkswagen Jetta when running at 2 600 rpm (maximum torque) corresponding to normal highway driving. Technical parameters for this engine are given on Page 9.

Consider a fuel-air cycle with modified specific heats and assume the inlet conditions to be ambient and atmospheric. Note that the mass of fuel must be included with the air (and gas) when calculating the cycle conditions. Use the following air-fuel characteristics:

Air-Fuel Ratio	r_f	=	15	
Calorific Value	CV	=	45 MJ/kg	
Specific Heat	c_p	=	1.42 kJ/kg°C	(air + fuel)
Specific Heat	c_v	=	1.14 kJ/kg°C	(air + fuel)
Inlet Pressure	p_1	=	100 kPa	
Inlet Temperature	t_1	=	30°C	

Note that only the temperatures at each point in the cycle need be calculated to determine the theoretical power.

- Sketch the cycle on a p-V diagram and label all key points in the cycle that will be used to identify calculated values. (1)
- Calculate the cylinder volume at the beginning and end of the compression stroke. (2)
- Calculate the mass of air and fuel ($m_{\text{air}} + m_{\text{fuel}}$) in the cylinder during the cycle and the mass of fuel (m_{fuel}) available for heat release. (2)
- Calculate the temperatures at all key points in the cycle. (3)
- Calculate the net work out from the cycle (kJ) (one cylinder only). (3)
- Calculate the ideal cycle power output (kW) for the whole engine at 2 600 rpm (1)
- Calculate the actual cycle power output (kW) taking account of process losses (time loss 6%, blowdown loss 2%, heat loss 12%) of 20%. (1)
- Calculate the useful power output (kW) taking account of mechanical losses (pumping loss 2%, friction loss 16%) of 18%. (1)
- Calculate the specified power (kW) at 2 600 rpm from the given specifications. Compare this with the power calculated in (h) and comment on the result. (1)

[15 marks]

QUESTION 4 STEAM PLANT HEAT REJECTION**PART I CONDENSER PERFORMANCE**

Refer to the Examination Paper Attachments Page 10 **Koeberg Condenser** and Page 11 **Temperature Profiles**. Note that 1 bar = 0.1 MPa.

Consider the condenser to be operating under the given conditions. Sketch, in dotted lines on each of the given axes, the design temperature profile, with specified temperatures for both cooling water and steam, along the condenser tubes (from inlet to outlet). Show clearly the change in cooling water temperature ΔT and the difference between the average cooling water temperature and the condensing steam temperature θ .

For the following no detailed calculations are required and temperatures should be rounded to the nearest 1°C . The estimates should be based on average temperature differences (not log mean temperature differences) and in each case the new values for ΔT and θ should be stated.

If the conditions are changed as indicated below, sketch, in solid lines on the given axes, the anticipated temperature profiles, with numerical values for both cooling water and steam, across the condenser for each of the following conditions:

- (a) Cooling water inlet temperature increased to 18°C . (1½)
- (b) Turbine load reduced to one quarter of its original value. (2½)
- (c) Cooling water flow reduced to one half of its original value which also results in the overall heat transfer coefficient being reduced to 70% of its original value. (2½)
- (d) Overall heat transfer coefficient reduced by 20% due to fouling of tubes. (2½)

(9 marks)

Question 4 continued on next page

Question 4 continued.

PART II COOLING TOWER

Refer to the Examination Paper Attachments Page 12 **Cooling Tower Evaporative Loss.**

A coal fired power plant with an electrical output of 600 MW rejects 1500 MJ/s of heat to the atmosphere via a steam condenser and a wet natural draught cooling tower. Operating conditions are as follows:

Steam inlet (turbine exhaust) temperature	30°C
Cooling water inlet temperature	15°C
Cooling water outlet temperature	25°C
Ambient air temperature	30°C.
Relative air humidity	40%

Determine the following:

- (a) Flow rate of cooling water (m^3/s). (2)
- (b) Evaporative loss in cooling tower (m^3/GJ). (1)
- (c) Evaporative loss in cooling tower (m^3/s). (1)
- (d) Percentage loss of cooling water (%). (1)
- (e) Consumption of water by cooling tower (L/kWh generated) (litres/unit generated). (1)

(6 marks)

[15 marks]

SECTION B DESCRIPTIVE QUESTIONS

Descriptive questions (but not graphical questions done on attachments) should be answered in essay form, with sketches if appropriate, and taking approximately one full page for every 5 marks. A full page means approximately 250 words unless diagrams take the place of some words.

QUESTION 5 BRAYTON CYCLE MODIFICATIONS

Refer to the Examination Paper Attachments Pages 13 and 14 **Brayton Cycle Modifications**.

Part (a) must be done on the attachments which must be returned with the examination booklet.

(a) For each of the following modifications to the basic cycle sketch, on a T-s diagram, the basic cycle and the modified cycle.

- (i) increased pressure ratio
- (ii) regenerative heating
- (iii) compressor intercooling
- (iv) turbine reheating
- (v) exhaust afterburning

In each case assume that the turbine inlet temperature is at its limiting (maximum) value (before and after the modification) and that the atmospheric air inlet temperature is constant.

(10 marks)

(b) State with reasons what the advantages and disadvantages are of each modified cycle and how the efficiency and power output is likely to be affected. Where appropriate give examples of practical applications of the cycles to support the choice of particular modifications

(5 marks)

[15 marks]

QUESTION 6 FUEL CHARACTERISTICS

PART I FOSSIL FUEL

- (a) State what is meant by heating value and clarify the difference between higher heating value and lower heating value. State which one is commonly used.
- (b) Compare and state the characteristics (constituents and heating value) of coals of different grade or rank. Indicate how and why these characteristics change according to the degree of transformation from vegetal matter to coal.
- (c) With regard to coal, state what constitutes a Proximate Analysis and what constitutes an Ultimate Analysis. Clarify the usefulness of each.

(9 marks)

PART II NUCLEAR FUEL

For a nuclear reactor of your choice:

- (a) Describe the nuclear fission process. Clarify what fuel is used, how fission is initiated and what components are produced. State the necessary characteristics of a nuclear fuel and describe the properties of a typical fuel.
- (b) Explain the design requirements of a nuclear reactor. Describe the main internal components and clarify what purpose they serve. Emphasis should be on how the chain reaction is maintained and how energy is produced and removed from the reactor core.

(6 marks)

[15 marks]

EXAMINATION PAPER ATTACHMENTS

QUESTION 3 VOLKSWAGEN JETTA 2003 MODEL

2.0 L GASOLINE ENGINE

Description	Specification
Engine	
Type	2.0 L, 4 cylinder, in-line
Bore	82.5 mm
Stroke	92.8 mm
Displacement	1,984 cm ³
Compression Ratio	10.0:1
Horsepower (SAE) @ rpm	115 @ 5,200 (85 kW @ 5,200)
Maximum torque, lbs - ft @ rpm	122 @ 2,600 (165 Nm @ 2,600)
Fuel Requirement	Regular unleaded
Firing Order	1-3-4-2
Engine Design	
Arrangement	Front mounted, transverse
Cylinder Block	Cast iron
Crank Shaft	Cast iron, five main bearings
Cylinder Head	Aluminum alloy, cross flow
Valve Train	Single overhead camshaft, spur belt driven with semi-automatic belt tensioner, two valves per cylinder, maintenance free hydraulic lifters, single coil valve springs
Cooling System	Water cooled, water pump, cross flow radiator, double thermostatically controlled electric 2-speed radiator fan
Lubrication	Rotary internal gear pump, intermediate shaft driven, oil cooler
Fuel / Air Supply	Sequential multi-point fuel injection (Motronic)
Emissions	Bin 3 EPA Federal Emissions Concept, ORVR (On-board Refuelling Vapor Recovery), EVAP (enhanced evaporation system) standards for USA, 3-way catalytic converter with two oxygen sensors (up - and downstream)

QUESTION 4 PART I KOEBERG CONDENSER

NAME

Steam flow rate	2996 t/h
Water make-up flow rate	9 t/h
Cooling water flow rate	141 000 t/h
Cooling water inlet temperature	13°C
Cooling water outlet temperature	24°C
Cooling water density	1.025
Cooling water friction head loss	4.7 m
Mean steam velocity at tube bank	92 m/s
Cooling water velocity inside tubes	2 m/s
Number of tubes	76968
Number of support plates	14 (per bundle)
Tube material	titanium
Cooling surface area	57 426 m ²
Tube overall length	12.84 m
Tube effective length	12.50 m
Tube diameter (OD)	19 mm
Tube wall thickness (normal tubes)	0.5 mm
Tube wall thickness (impact tubes)	0.6 mm
Tube configuration	diagonal array
Tube pitch across array	26 mm
Tube pitch along array	45 mm
Tube fixing method	expanding
Tube mass	132 t
Total volume under vacuum	7500 m ³
Steam inlet pressure	0.043 bar abs
Steam inlet temperature	30°C
Terminal temperature difference	6°C
Condenser hotwell capacity	700 m ³ (approx.)
Number of water boxes (inlet and outlet)	12
Water box internal lining	neoprene
Condenser shell thickness	18 mm
Tube plate thickness	25 mm
Support plate thickness	12 mm
Condenser length	43 m (approx.)
Condenser width	25 m (approx.)
Condenser mass without LP Heaters	1267 t

QUESTION 4 PART I TEMPERATURE PROFILES

NAME

Show initial conditions as dotted lines on each diagram

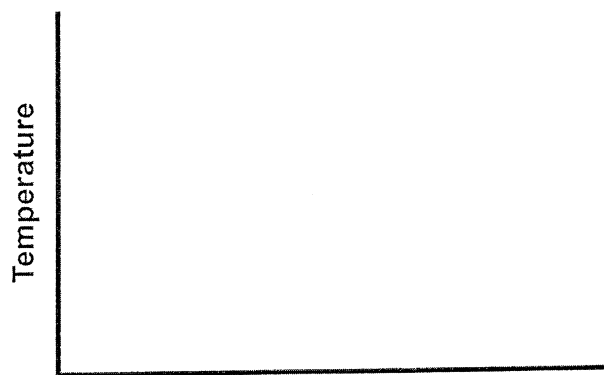
Show new conditions for each case as solid lines

Give temperatures on axes

Show basic calculations and new values for ΔT and θ below each diagram

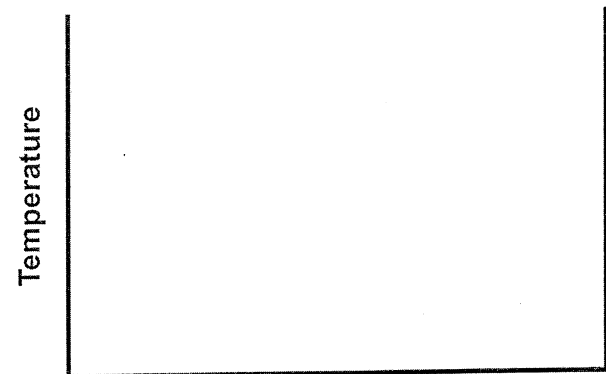
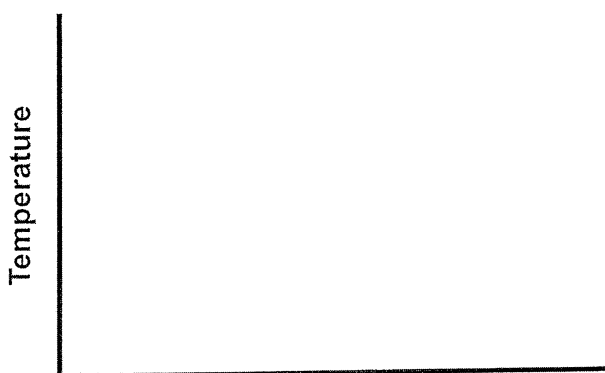
(a) Increase in cooling water temperature

(b) Reduction in turbine load



(c) Reduction in cooling water flow

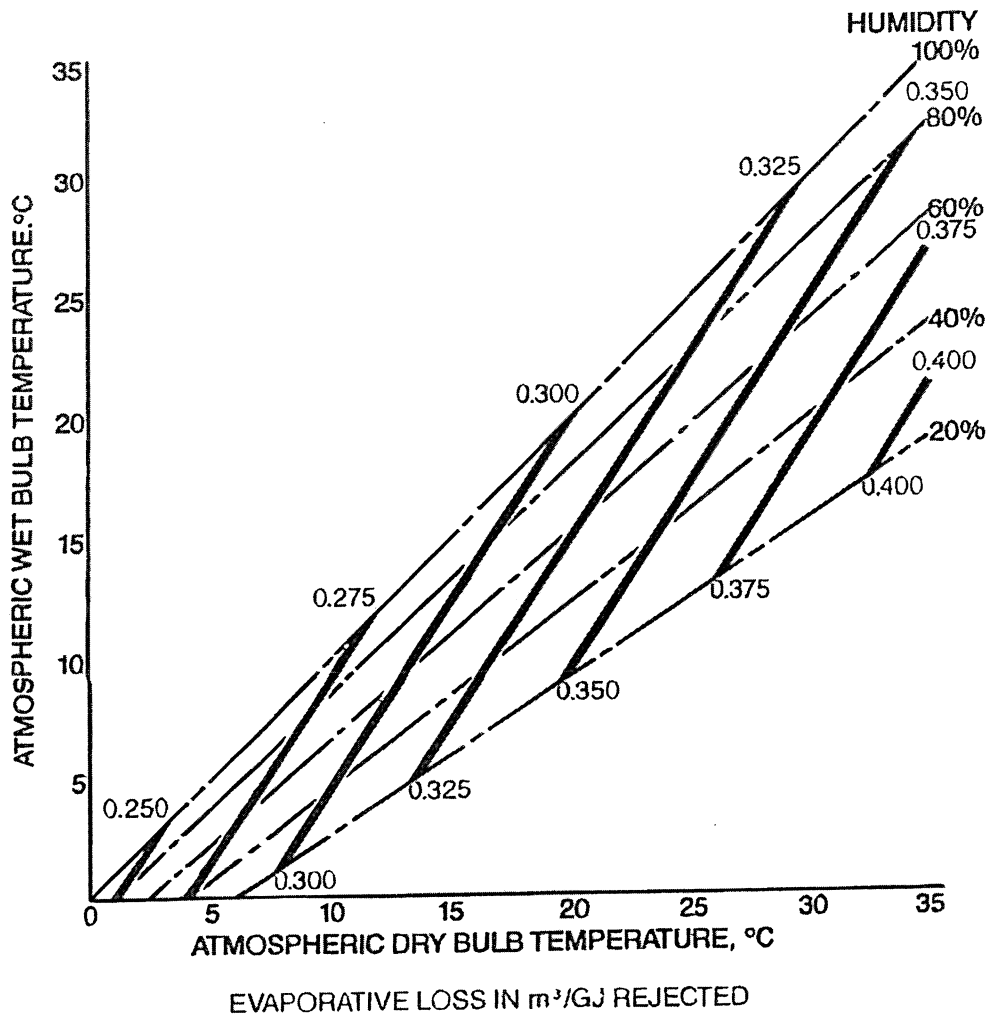
(d) Reduction in heat transfer coefficient



EXAMINATION PAPER ATTACHMENTS

NAME

QUESTION 4 PART II COOLING TOWER EVAPORATION LOSS



Evaporative loss from natural draught cooling towers

The chart is used to estimate the evaporative loss in m^3/GJ of heat rejected.

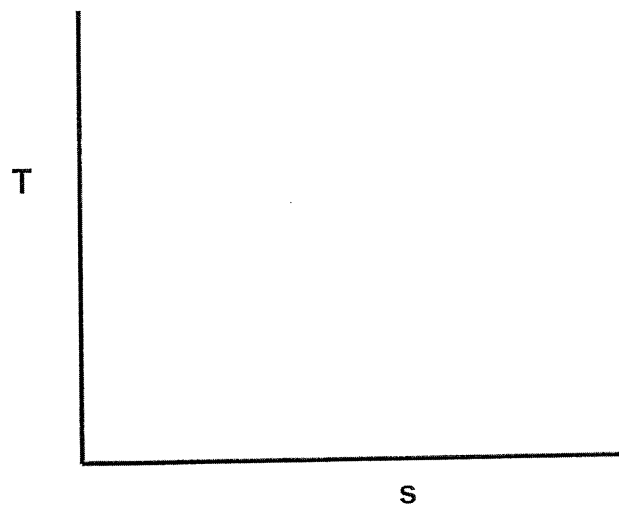
EXAMINATION PAPER ATTACHMENTS

NAME

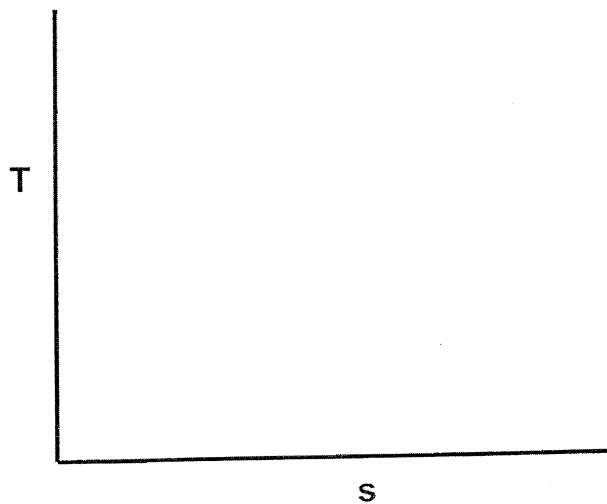
QUESTION 5 BRAYTON CYCLE MODIFICATIONS

(a) On each T-s diagram sketch a basic Brayton Cycle and show how it is modified in each case (with fixed compressor and turbine inlet temperatures).

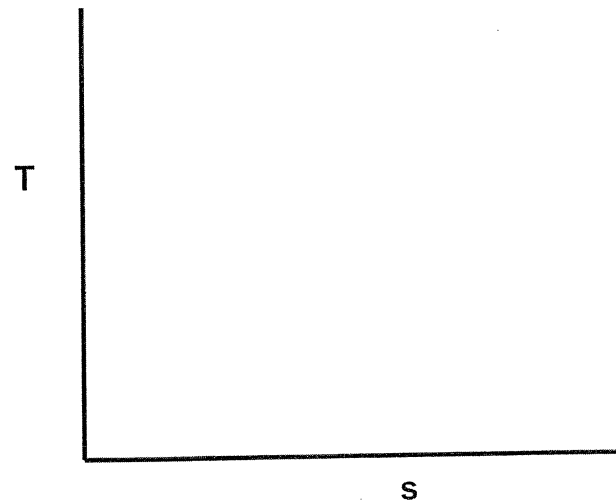
(i) Increased Pressure Ratio



(ii) Regenerative Heating



(iii) Compressor Intercooling



EXAMINATION PAPER ATTACHMENTS

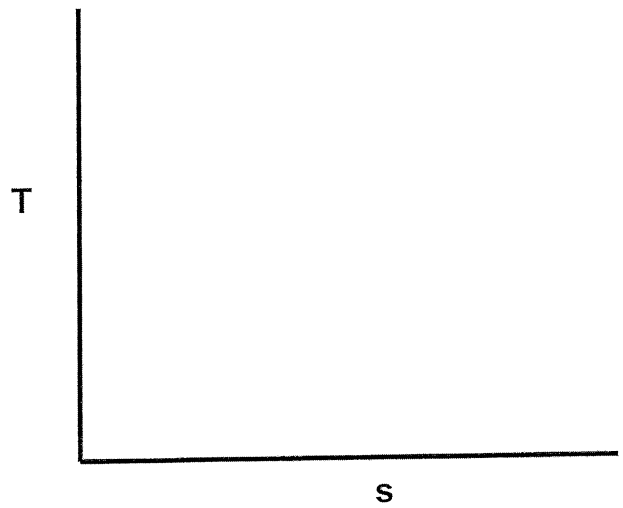
NAME

QUESTION 4 CONTINUED

(iv) Turbine Reheating



(v) Exhaust Afterburning



EXAMINATION REFERENCE MATERIAL

NOMENCLATURE FOR REFERENCE EQUATIONS (SI UNITS)

a	Acceleration	m/s^2
A	Flow area, Surface area	m^2
c_p	Specific heat at constant pressure	$J/kg^\circ C$
c_v	Specific heat at constant volume	$J/kg^\circ C$
D	Diameter	m
E	Energy	J
E_f	Energy release per fission of one atom	
h	Specific enthalpy	J/kg
H	Enthalpy	J
F	Force	N
g	Gravitational acceleration	m/s^2
k	Ratio of specific heats	
L	Length	m
m	Mass	kg
\dot{m}	Fractional mass flow rate	
M	Mass flow rate	kg/s
M	Molecular weight	
N	Number of nuclei	number/g
N_A	Avogadro's Number	
N_f	Number of fissile nuclei	number/cm ³
n	Gas expansion index	
p	Pressure	Pa
P	Power	W
q	Heat transferred	J/kg
q^*	Heat release rate	J/cm^3
Q	Heat	J
Q	Volume flow rate	m^3/s
R	Specific gas constant	$J/kg^\circ K$
R_0	Universal gas constant	$J/kg\text{-mole}^\circ K$
s	Specific entropy	$J/kg^\circ K$
S	Entropy	$J/^\circ K$
t	Time	s
t	Temperature	$^\circ C$
T	Absolute temperature	$^\circ K$
u	Specific internal energy	J/kg
U	Internal energy	J
v	Specific volume	m^3/kg
V	Volume	m^3
V	Velocity	m/s
w	Specific work	J/kg
W	Work	J

x	Length	m
z	Elevation	m
γ	Fuel enrichment	
η	Efficiency	
ϕ	Neutron flux	neutrons/cm ² s
σ_f	Cross section	barn
μ	Dynamic viscosity	Ns/m ²
ν	Kinematic viscosity	m ² /s
ρ	Density	kg/m ³
τ	Thrust	N
τ	Torque	Nm
Ω	Heat transfer rate	J/s

CONSTANTS

For consistency in calculations the following constants should be used:

Gravitational Acceleration	$g = 9.81 \text{ m/s}$
Atmospheric Pressure	$p = 100 \text{ kPa}$
Universal Gas Constant	$R_o = 8.314 \text{ kJ/kg mole}^\circ\text{K}$
Density of Water	$\rho = 1000 \text{ kg/m}^3$
Specific Heat of Water	$c_p = 4.19 \text{ kJ/kg}^\circ\text{C}$
Specific Heat of Air	$c_p = 1.005 \text{ kJ/kg}^\circ\text{C}$
Specific Heat of Air	$c_v = 0.718 \text{ kJ/kg}^\circ\text{C}$
Specific Heat of Helium	$c_p = 5.193 \text{ kJ/kg}^\circ\text{C}$
Specific Heat of Helium	$c_v = 3.116 \text{ kJ/kg}^\circ\text{C}$
Specific Gas Constant for Air	$R = 0.287 \text{ kJ/kg}^\circ\text{K}$
Avogadro's Number	$N_A = 0.602 \times 10^{24} \text{ atoms/mole}$
Nuclear Cross Section	$1 \text{ barn} = 10^{-24} \text{ cm}^2$

GENERAL REFERENCE EQUATIONS

Ideal Gas Relationships

Gas Law:	$pv = RT$
Gas Law:	$pV = mRT$
Specific Heat at Constant Pressure:	$c_p = \Delta h/\Delta T$
Specific Heat at Constant Volume:	$c_v = \Delta u/\Delta T$
Gas Constant:	$R = c_p - c_v$
Specific Heat Ratio:	$k = c_p/c_v$

Constant Volume:
 Constant Pressure:
 Constant Temperature:
 Constant Entropy:
 Isentropic Relations:

$$T_1/T_2 = p_1/p_2$$

$$T_1/T_2 = v_1/v_2$$

$$p_1 v_1 = p_2 v_2$$

$$p_1 v_1^k = p_2 v_2^k$$

$$p_1/p_2 = (v_2/v_1)^k = (T_1/T_2)^{k/(k-1)}$$

$$T_1/T_2 = (v_2/v_1)^{k-1} = (p_1/p_2)^{(k-1)/k}$$

Work in Non-Flow Processes

Constant Pressure:
 Constant Temperature:
 Constant Entropy:

$$w = p (v_2 - v_1)$$

$$w = p_1 v_1 \ln(v_2/v_1)$$

$$w = (p_2 v_2 - p_1 v_1) / (1 - k)$$

$$w = (T_2 - T_1) R / (1 - k)$$

Work in Flow Processes

Constant Temperature:
 Constant Volume:
 Constant Entropy:

$$w = p_1 v_1 \ln(v_2/v_1)$$

$$w = (p_2 - p_1) v$$

$$w = (p_1 v_1 - p_2 v_2) k / (k - 1)$$

Thermodynamics

First Law:
 Enthalpy:
 Enthalpy Change
 Continuity:
 Flow Work:
 Energy Equation:
 Entropy:

$$dE = \delta Q - \delta W$$

$$h = u + pv$$

$$\Delta h = \Delta u + \Delta(pv)$$

$$\rho VA = \text{constant}$$

$$w = \Delta(pv)$$

$$zg + V^2/2 + u + pv + \Delta w + \Delta q = \text{constant}$$

$$\Delta s = q/T \quad (\text{reversible conditions})$$

Fluid Mechanics

Continuity Equation:
 Energy Equation:
 Bernoulli's Equation:
 Momentum Equation:

$$\rho_1 V_1 A_1 = \rho_2 V_2 A_2 = M$$

$$z_1 g + V_1^2/2 + u_1 + p_1 v_1 + w_{in} + q_{in}$$

$$= z_2 g + V_2^2/2 + u_2 + p_2 v_2 + w_{out} + q_{out}$$

$$p_1/\rho g + z_1 + V_1^2/2g = p_2/\rho g + z_2 + V_2^2/2g$$

$$F = p_1 A_1 - p_2 A_2 - \rho VA(V_2 - V_1)$$

(one dimensional)

Internal Combustion Engines

Power Output
 Engine Capacity
 Mean Effective Pressure

$$P = 2\pi N\tau / 60$$

$$V_{total} = 1000 (\pi D^2/4) LN_{cylinders}$$

$$MEP = \text{Work} / (V_1 - V_2)$$

Steam Turbines

Nozzle Equation: $h_1 - h_2 = (V_2^2 - V_1^2) / 2$
 Work: $W = [(V_1^2_{\text{absolute}} - V_2^2_{\text{absolute}}) + (V_2^2_{\text{relative}} - V_1^2_{\text{relative}})] / 2$

Gas Turbines

Isentropic Equation: $(T_2/T_1) = (p_2/p_1)^{(k-1)/k}$
 Enthalpy Change: $h_1 - h_2 = c_p(T_1 - T_2)$ (ideal gas)
 Nozzle Equation: $h_1 - h_2 = (V_2^2 - V_1^2) / 2$

Jet Propulsion

Thrust: $T = M(V_{\text{jet}} - V_{\text{aircraft}})$
 Thrust Power: $T V_{\text{aircraft}} = M(V_{\text{jet}} - V_{\text{aircraft}}) V_{\text{aircraft}}$
 Jet Power: $P = M(V_{\text{jet}}^2 - V_{\text{aircraft}}^2) / 2$
 Propulsion Efficiency: $\eta_p = 2V_{\text{aircraft}} / (V_{\text{jet}} + V_{\text{aircraft}})$

Wind Turbines

Maximum Ideal Power: $P_{\text{max}} = 8 \rho A V_1^3 / 27$

Nuclear Energy

Number of nuclei per gram of material: $N = N_A / M$
 Number of fissile nuclei per cm³ of material: $N_f = \gamma (N_A / M) \rho$
 Heat release rate in nuclear fuel: $q^* = \phi N_f \sigma_f E_f$

Cycle Efficiencies

$\eta_{\text{cycle}} = W_{\text{out}} / q_{\text{in}} = W_{\text{out}} / Q_{\text{in}} = P_{\text{out}} / \Omega_{\text{in}}$
 $\eta_{\text{Carnot}} = (T_{\text{hot}} - T_{\text{cold}}) / T_{\text{hot}}$
 $\eta_{\text{Rankine}} = (\Delta h_{\text{turbine}} - \Delta h_{\text{pump}}) / \Delta h_{\text{boiler}}$
 $\eta_{\text{Brayton}} = (\Delta T_{\text{turbine}} - \Delta T_{\text{Compressor}}) / \Delta T_{\text{combustion}}$

Component Efficiencies

$\eta_{\text{boiler}} = \Omega_{\text{out}} / \Omega_{\text{in}}$
 $\eta_{\text{boiler}} = (\Omega_{\text{in}} / \Omega_{\text{lost}}) / \Omega_{\text{in}}$
 $\eta_{\text{turbine}} = \Delta h_{\text{actual}} / \Delta h_{\text{isentropic}}$
 $\eta_{\text{nozzle}} = \Delta h_{\text{actual}} / \Delta h_{\text{isentropic}}$
 $\eta_{\text{gas turbine}} = \Delta T_{\text{actual}} / \Delta T_{\text{isentropic}}$
 $\eta_{\text{pump}} = \Delta h_{\text{isentropic}} / \Delta h_{\text{actual}}$
 $\eta_{\text{compressor}} = \Delta T_{\text{isentropic}} / \Delta T_{\text{actual}}$

Thermodynamics and Heat Power

SIXTH EDITION

Irving Granet, P.E.

late, Queensborough Community College of City University of New York

Maurice Bluestein, Ph.D.

Indiana University—Purdue University, Indianapolis

PRENTICE HALL

Upper Saddle River, New Jersey Columbus, Ohio

TABLE A.1 (SI)

Saturation: Temperature (Steam)

Temp. °C T	Press. kPa P	Specific Volume (m ³ /kg)		Internal Energy (kJ/kg)			Enthalpy (kJ/kg)			Entropy (kJ/kg·°K)		
		Sat. Liquid v_f	Sat. Vapor v_g	Sat. Liquid u_f	Evap. u_{fg}	Sat. Vapor u_g	Sat. Liquid h_f	Evap. h_{fg}	Sat. Vapor h_g	Sat. Liquid s_f	Evap. s_{fg}	Sat. Vapor s_g
0.01	0.6113	0.001 000	206.14	.00	2375.3	2375.3	.01	2501.3	2501.4	.0000	9.1562	9.1562
5	0.8721	0.001 000	147.12	20.97	2361.3	2382.3	20.98	2489.6	2510.6	.0761	8.9496	9.0257
10	1.2276	0.001 000	106.38	42.00	2347.2	2389.2	42.01	2477.7	2519.8	.1510	8.7498	8.9008
15	1.7051	0.001 001	77.93	62.99	2333.1	2396.1	62.99	2465.9	2528.9	.2245	8.5569	8.7814
20	2.339	0.001 002	57.79	83.95	2319.0	2402.9	83.96	2454.1	2538.1	.2966	8.3706	8.6672
25	3.169	0.001 003	43.36	104.88	2304.9	2409.8	104.89	2442.3	2547.2	.3674	8.1905	8.5580
30	4.246	0.001 004	32.89	125.78	2290.8	2416.6	125.79	2430.5	2556.3	.4369	8.0164	8.4533
35	5.628	0.001 006	25.22	146.67	2276.7	2423.4	146.68	2418.6	2565.3	.5053	7.8478	8.3531
40	7.384	0.001 008	19.52	167.56	2262.6	2430.1	167.57	2406.7	2574.3	.5725	7.6845	8.2570
45	9.593	0.001 010	15.26	188.44	2248.4	2436.8	188.45	2394.8	2583.2	.6387	7.5261	8.1648
50	12.349	0.001 012	12.03	209.32	2234.2	2443.5	209.33	2382.7	2592.1	.7038	7.3725	8.0763
55	15.758	0.001 015	9.568	230.21	2219.9	2450.1	230.23	2370.7	2600.9	.7679	7.2234	7.9913
60	19.940	0.001 017	7.671	251.11	2205.5	2456.6	251.13	2358.5	2609.6	.8312	7.0784	7.9096
65	25.03	0.001 020	6.197	272.02	2191.1	2463.1	272.06	2346.2	2618.3	.8935	6.9375	7.8310
70	31.19	0.001 023	5.042	292.95	2176.6	2469.6	292.98	2333.8	2626.8	.9549	6.8004	7.7553
75	38.58	0.001 026	4.131	313.90	2162.0	2475.9	313.93	2321.4	2635.3	1.0155	6.6669	7.6824
80	47.39	0.001 029	3.407	334.86	2147.4	2482.2	334.91	2308.8	2643.7	1.0753	6.5369	7.6122
85	57.83	0.001 033	2.828	355.84	2132.6	2488.4	355.90	2296.0	2651.9	1.1343	6.4102	7.5445
90	70.14	0.001 036	2.361	376.85	2117.7	2494.5	376.92	2283.2	2660.1	1.1925	6.2866	7.4791
95	84.55	0.001 040	1.982	397.88	2102.7	2500.6	397.96	2270.2	2668.1	1.2500	6.1659	7.4159

TABLE A.1 (SI) (cont'd.)

Temp. °C T	Press. kPa P	Specific Volume (m ³ /kg)		Internal Energy (kJ/kg)			Enthalpy (kJ/kg)			Entropy (kJ/kg·°K)		
		Sat. Liquid v_f	Sat. Vapor v_g	Sat. Liquid u_f	Sat. Vapor u_g	Sat. Liquid h_f	Sat. Vapor h_g	Sat. Liquid s_f	Sat. Vapor s_g	Sat. Liquid s_f	Sat. Vapor s_g	
100	0.101 35	0.001 044	1.6729	418.94	2087.6	419.04	2257.0	2676.1	1.3069	6.0480	7.3549	
105	0.120 82	0.001 048	1.4194	440.02	2072.3	440.15	2243.7	2683.8	1.3630	5.9328	7.2958	
110	0.143 27	0.001 052	1.2102	461.14	2057.0	461.30	2230.2	2691.5	1.4185	5.8202	7.2387	
115	0.169 06	0.001 056	1.0366	482.30	2041.4	482.48	2216.5	2699.0	1.4734	5.7100	7.1833	
120	0.198 53	0.001 060	0.8919	503.50	2025.8	503.71	2202.6	2706.3	1.5276	5.6020	7.1296	
125	0.2321	0.001 065	0.7706	524.74	2009.9	524.99	2188.5	2713.5	1.5813	5.4962	7.0775	
130	0.2701	0.001 070	0.6685	546.02	1993.9	546.31	2174.2	2720.5	1.6344	5.3925	7.0269	
135	0.3130	0.001 075	0.5822	567.35	1977.7	567.69	2159.6	2727.3	1.6870	5.2907	6.9777	
140	0.3613	0.001 080	0.5089	588.74	1961.3	589.13	2144.7	2733.9	1.7391	5.1908	6.9299	
145	0.4154	0.001 085	0.4463	610.18	1944.7	610.63	2129.6	2740.3	1.7907	5.0926	6.8833	
150	0.4758	0.001 091	0.3928	631.68	1927.9	632.20	2114.3	2746.5	1.8418	4.9960	6.8379	
155	0.5431	0.001 096	0.3468	653.24	1910.8	653.84	2098.6	2752.4	1.8925	4.9010	6.7935	
160	0.6178	0.001 102	0.3071	674.87	1893.5	675.55	2082.6	2758.1	1.9427	4.8075	6.7502	
165	0.7005	0.001 108	0.2727	696.56	1876.0	697.34	2066.2	2763.5	1.9925	4.7153	6.7078	
170	0.7917	0.001 114	0.2428	718.33	1858.1	719.21	2049.5	2768.7	2.0419	4.6244	6.6663	
175	0.8920	0.001 121	0.2168	740.17	1840.0	741.17	2032.4	2773.6	2.0909	4.5347	6.6256	
180	1.0021	0.001 127	0.194 05	762.09	1821.6	763.22	2015.0	2778.2	2.1396	4.4461	6.5857	
185	1.1227	0.001 134	0.174 09	784.10	1802.9	785.37	1997.1	2782.4	2.1879	4.3586	6.5465	
190	1.2544	0.001 141	0.156 54	806.19	1783.8	807.62	1978.8	2786.4	2.2359	4.2720	6.5079	
195	1.3978	0.001 149	0.141 05	828.37	1764.4	829.98	1960.0	2790.0	2.2835	4.1863	6.4698	
200	1.5538	0.001 157	0.127 36	850.65	1744.7	852.45	1940.7	2793.2	2.3309	4.1014	6.4323	
205	1.7230	0.001 164	0.115 21	873.04	1724.5	875.04	1921.0	2796.0	2.3780	4.0172	6.3952	
210	1.9062	0.001 173	0.104 41	895.53	1703.9	897.76	1900.7	2798.5	2.4248	3.9337	6.3585	
215	2.104	0.001 181	0.094 79	918.14	1682.9	920.62	1879.9	2800.5	2.4714	3.8507	6.3221	
220	2.318	0.001 190	0.086 19	940.87	1661.5	943.62	1858.5	2802.1	2.5178	3.7683	6.2861	
225	2.548	0.001 199	0.078 49	963.73	1639.6	966.78	1836.5	2803.3	2.5639	3.6863	6.2503	
230	2.795	0.001 209	0.071 58	986.74	1617.2	990.12	1813.8	2804.0	2.6099	3.6047	6.2146	
235	3.060	0.001 219	0.065 37	1009.89	1594.2	1013.62	1790.5	2804.2	2.6558	3.5233	6.1791	
240	3.344	0.001 229	0.059 76	1033.21	1570.8	1037.32	1766.5	2803.8	2.7015	3.4422	6.1437	
245	3.648	0.001 240	0.054 71	1056.71	1546.7	1061.23	1741.7	2803.0	2.7472	3.3612	6.1083	

TABLE A.1 (SI) (cont'd.)

Temp. °C T	Press. MPa P	Specific Volume (m³/kg)				Internal Energy (kJ/kg)				Enthalpy (kJ/kg)				Entropy (kJ/kg · °K)			
		Sat. Liquid v_f	Sat. Vapor v_g	Sat. Liquid u_f	Sat. Vapor u_g	Evap. u_{fg}	Sat. Vapor u_g	Sat. Liquid h_f	Sat. Vapor h_g	Evap. h_{fg}	Sat. Vapor h_g	Sat. Liquid s_f	Sat. Vapor s_g	Evap. s_{fg}	Sat. Vapor s_g		
250	3.973	0.001 251	0.050 13	1080.39	1522.0	2602.4	1085.36	2801.5	1716.2	2792.7	2.7927	3.2802	3.2802	6.0730			
255	4.319	0.001 263	0.045 98	1104.28	1496.7	2600.9	1109.73	2799.5	1689.8	2.8383	3.1992	3.1992	6.0375				
260	4.688	0.001 276	0.042 21	1128.39	1470.6	2599.0	1134.37	2796.9	1662.5	2.8838	3.1181	3.1181	6.0019				
265	5.081	0.001 289	0.038 77	1152.74	1443.9	2596.6	1159.28	2793.6	1634.4	2.9294	3.0368	3.0368	5.9662				
270	5.499	0.001 302	0.035 64	1177.36	1416.3	2593.7	1184.51	2789.7	1605.2	2.9751	2.9551	2.9551	5.9301				
275	5.942	0.001 317	0.032 79	1202.25	1387.9	2590.2	1210.07	2785.0	1574.9	3.0208	2.8730	2.8730	5.8938				
280	6.412	0.001 332	0.030 17	1227.46	1358.7	2586.1	1235.99	2779.6	1543.6	3.0668	2.7903	2.7903	5.8571				
285	6.909	0.001 348	0.027 77	1253.00	1328.4	2581.4	1262.31	2773.3	1511.0	3.1130	2.7070	2.7070	5.8199				
290	7.436	0.001 366	0.025 57	1278.92	1297.1	2576.0	1289.07	2766.2	1477.1	3.1594	2.6227	2.6227	5.7821				
295	7.993	0.001 384	0.023 54	1305.2	1264.7	2569.9	1316.3	2758.1	1441.8	3.2062	2.5375	2.5375	5.7437				
300	8.581	0.001 404	0.021 67	1332.0	1231.0	2563.0	1344.0	2749.0	1404.9	3.2534	2.4511	2.4511	5.7045				
305	9.202	0.001 425	0.019 948	1359.3	1195.9	2555.2	1372.4	2738.7	1366.4	3.3010	2.3633	2.3633	5.6643				
310	9.856	0.001 447	0.018 350	1387.1	1159.4	2546.4	1401.3	2727.3	1326.0	3.3493	2.2737	2.2737	5.6230				
315	10.547	0.001 472	0.016 867	1415.5	1121.1	2536.6	1431.0	2714.5	1283.5	3.3982	2.1821	2.1821	5.5804				
320	11.274	0.001 499	0.015 488	1444.6	1080.9	2525.5	1461.5	2700.1	1238.6	3.4480	2.0882	2.0882	5.5362				
330	12.845	0.001 561	0.012 996	1505.3	993.7	2498.9	1525.3	2665.9	1140.6	3.5507	1.8909	1.8909	5.4417				
340	14.586	0.001 638	0.010 797	1570.3	894.3	2464.6	1594.2	2622.0	1027.9	3.6594	1.6763	1.6763	5.3357				
350	16.513	0.001 740	0.008 813	1641.9	776.6	2418.4	1670.6	2563.9	893.4	3.7777	1.4335	1.4335	5.2112				
360	18.651	0.001 893	0.006 945	1725.2	626.3	2351.5	1760.5	2481.0	720.5	3.9147	1.1379	1.1379	5.0526				
370	21.03	0.002 213	0.004 925	1844.0	384.5	2228.5	1890.5	2332.1	441.6	4.1106	.6865	.6865	4.7971				
374.14	22.09	0.003 155	0.003 155	2029.6	0	2029.6	2099.3	0	0	4.4298	0	0	4.4298				

TABLE A.2 (SI)
Saturation Pressures (Steam)

Press. kPa <i>P</i>	Temp. °C <i>T</i>	Specific Volume (m ³ /kg)		Internal Energy (kJ/kg)			Enthalpy (kJ/kg)			Entropy (kJ/kg · °K)		
		Sat. Liquid <i>v_f</i>	Sat. Vapor <i>v_g</i>	Sat. Liquid <i>u_f</i>	Evap. <i>u_{fg}</i>	Sat. Vapor <i>u_g</i>	Sat. Liquid <i>h_f</i>	Evap. <i>h_{fg}</i>	Sat. Vapor <i>h_g</i>	Sat. Liquid <i>s_f</i>	Evap. <i>s_{fg}</i>	Sat. Vapor <i>s_g</i>
0.6113	0.01	0.001 000	206.14	.00	2375.3	2375.3	.01	2501.3	2501.4	.0000	9.1562	9.1562
1.0	6.98	0.001 000	129.21	29.30	2355.7	2385.0	29.30	2484.9	2514.2	.1059	8.8697	8.9756
1.5	13.03	0.001 001	87.98	54.71	2338.6	2393.3	54.71	2470.6	2525.3	.1957	8.6322	8.8279
2.0	17.50	0.001 001	67.00	73.48	2326.0	2399.5	73.48	2460.0	2533.5	.2607	8.4629	8.7237
2.5	21.08	0.001 002	54.25	88.48	2315.9	2404.4	88.49	2451.6	2540.0	.3120	8.3311	8.6432
3.0	24.08	0.001 003	45.67	101.04	2307.5	2408.5	101.05	2444.5	2545.5	.3545	8.2231	8.5776
4.0	28.96	0.001 004	34.80	121.45	2293.7	2415.2	121.46	2432.9	2554.4	.4226	8.0520	8.4746
5.0	32.88	0.001 005	28.19	137.81	2282.7	2420.5	137.82	2423.7	2561.5	.4764	7.9187	8.3951
7.5	40.29	0.001 008	19.24	168.78	2261.7	2430.5	168.79	2406.0	2574.8	.5764	7.6750	8.2515
10	45.81	0.001 010	14.67	191.82	2246.1	2437.9	191.83	2392.8	2584.7	.6493	7.5009	8.1502
15	53.97	0.001 014	10.02	225.92	2222.8	2448.7	225.94	2373.1	2599.1	.7549	7.2536	8.0085
20	60.06	0.001 017	7.649	251.38	2205.4	2456.7	251.40	2358.3	2609.7	.8320	7.0766	7.9085
25	64.97	0.001 020	6.204	271.90	2191.2	2463.1	271.93	2346.3	2618.2	.8931	6.9383	7.8314
30	69.10	0.001 022	5.229	289.20	2179.2	2468.4	289.23	2336.1	2625.3	.9439	6.8247	7.7686
40	75.87	0.001 027	3.993	317.53	2159.5	2477.0	317.58	2319.2	2636.8	1.0259	6.6441	7.6700
50	81.33	0.001 030	3.240	340.44	2143.4	2483.9	340.49	2305.4	2645.9	1.0910	6.5029	7.5939
75	91.78	0.001 037	2.217	384.31	2112.4	2496.7	384.39	2278.6	2663.0	1.2130	6.2434	7.4564
MPa												
0.100	99.63	0.001 043	1.6940	417.36	2088.7	2506.1	417.46	2258.0	2675.5	1.3026	6.0568	7.3594
0.125	105.99	0.001 048	1.3749	444.19	2069.3	2513.5	444.32	2241.0	2685.4	1.3740	5.9104	7.2844
0.150	111.37	0.001 053	1.1593	466.94	2052.7	2519.7	467.11	2226.5	2693.6	1.4336	5.7897	7.2233
0.175	116.06	0.001 057	1.0036	486.80	2038.1	2524.9	486.99	2213.6	2700.6	1.4849	5.6868	7.1717
0.200	120.23	0.001 061	0.8857	504.49	2025.0	2529.5	504.70	2201.9	2706.7	1.5301	5.5970	7.1271
0.225	124.00	0.001 064	0.7933	520.47	2013.1	2533.6	520.72	2191.3	2712.1	1.5706	5.5173	7.0878

TABLE A.2 (SI) (cont'd.)

Press. MPa <i>P</i>	Temp. °C <i>T</i>	Specific Volume		Internal Energy			Enthalpy			Entropy		
		Sat. Liquid <i>v_f</i>	Sat. Vapor <i>v_g</i>	Sat. Liquid <i>u_f</i>	Evap. <i>u_{fg}</i>	Sat. Vapor <i>u_g</i>	Sat. Liquid <i>h_f</i>	Evap. <i>h_{fg}</i>	Sat. Vapor <i>h_g</i>	Sat. Liquid <i>s_f</i>	Evap. <i>s_{fg}</i>	Sat. Vapor <i>s_g</i>
0.250	127.44	0.001 067	0.7187	535.10	2002.1	2537.2	535.37	2181.5	2716.9	1.6072	5.4455	7.0527
0.275	130.60	0.001 070	0.6573	548.59	1991.9	2540.5	548.89	2172.4	2721.3	1.6408	5.3801	7.0209
0.300	133.55	0.001 073	0.6058	561.15	1982.4	2543.6	561.47	2163.8	2725.3	1.6718	5.3201	6.9919
0.325	136.30	0.001 076	0.5620	572.90	1973.5	2546.4	573.25	2155.8	2729.0	1.7006	5.2646	6.9652
0.350	138.88	0.001 079	0.5243	583.95	1965.0	2548.9	584.33	2148.1	2732.4	1.7275	5.2130	6.9405
0.375	141.32	0.001 081	0.4914	594.40	1956.9	2551.3	594.81	2140.8	2735.6	1.7528	5.1647	6.9175
0.40	143.63	0.001 084	0.4625	604.31	1949.3	2553.6	604.74	2133.8	2738.6	1.7766	5.1193	6.8959
0.45	147.93	0.001 088	0.4140	622.77	1934.9	2557.6	623.25	2120.7	2743.9	1.8207	5.0359	6.8565
0.50	151.86	0.001 093	0.3749	639.68	1921.6	2561.2	640.23	2108.5	2748.7	1.8607	4.9606	6.8213
0.55	155.48	0.001 097	0.3427	655.32	1909.2	2564.5	655.93	2097.0	2753.0	1.8973	4.8920	6.7893
0.60	158.85	0.001 101	0.3157	669.90	1897.5	2567.4	670.56	2086.3	2756.8	1.9312	4.8288	6.7600
0.65	162.01	0.001 104	0.2927	683.56	1886.5	2570.1	684.28	2076.0	2760.3	1.9627	4.7703	6.7331
0.70	164.97	0.001 108	0.2729	696.44	1876.1	2572.5	697.22	2066.3	2763.5	1.9922	4.7158	6.7080
0.75	167.78	0.001 112	0.2556	708.64	1866.1	2574.7	709.47	2057.0	2766.4	2.0200	4.6647	6.6847
0.80	170.43	0.001 115	0.2404	720.22	1856.6	2576.8	721.11	2048.0	2769.1	2.0462	4.6166	6.6628
0.85	172.96	0.001 118	0.2270	731.27	1847.4	2578.7	732.22	2039.4	2771.6	2.0710	4.5711	6.6421
0.90	175.38	0.001 121	0.2150	741.83	1838.6	2580.5	742.83	2031.1	2773.9	2.0946	4.5280	6.6226
0.95	177.69	0.001 124	0.2042	751.95	1830.2	2582.1	753.02	2023.1	2776.1	2.1172	4.4869	6.6041
1.00	179.91	0.001 127	0.194 44	761.68	1822.0	2583.6	762.81	2015.3	2778.1	2.1387	4.4478	6.5865
1.10	184.09	0.001 133	0.177 53	780.09	1806.3	2586.4	781.34	2000.4	2781.7	2.1792	4.3744	6.5536
1.20	187.99	0.001 139	0.163 33	797.29	1791.5	2588.8	798.65	1986.2	2784.8	2.2166	4.3067	6.5233
1.30	191.64	0.001 144	0.151 25	813.44	1777.5	2591.0	814.93	1972.7	2787.6	2.2515	4.2438	6.4953
1.40	195.07	0.001 149	0.140 84	828.70	1764.1	2592.8	830.30	1959.7	2790.0	2.2842	4.1850	6.4693

TABLE A.2 (SI) (cont'd.)

Press. MPa <i>P</i>	Temp. °C <i>T</i>	Specific Volume (m ³ /kg)				Internal Energy (kJ/kg)				Enthalpy (kJ/kg)				Entropy (kJ/kg · °K)			
		Sat. Liquid <i>v_f</i>	Sat. Vapor <i>v_g</i>	Sat. Liquid <i>u_f</i>	Sat. Vapor <i>u_g</i>	Sat. Liquid <i>u_f</i>	Sat. Vapor <i>u_g</i>	Sat. Liquid <i>h_f</i>	Sat. Vapor <i>h_g</i>	Sat. Liquid <i>s_f</i>	Sat. Vapor <i>s_g</i>	Sat. Liquid <i>h_f</i>	Sat. Vapor <i>h_g</i>	Sat. Liquid <i>s_f</i>	Sat. Vapor <i>s_g</i>		
1.50	198.32	0.001 154	0.131 77	843.16	1751.3	2594.5	844.89	1947.3	2792.2	2.3150	4.1298	6.4448					
1.75	205.76	0.001 166	0.113 49	876.46	1721.4	2597.8	878.50	1917.9	2796.4	2.3851	4.0044	6.3896					
2.00	212.42	0.001 177	0.099 63	906.44	1693.8	2600.3	908.79	1890.7	2799.5	2.4474	3.8935	6.3409					
2.25	218.45	0.001 187	0.088 75	933.83	1668.2	2602.0	936.49	1865.2	2801.7	2.5035	3.7937	6.2972					
2.5	223.99	0.001 197	0.079 98	959.11	1644.0	2603.1	962.11	1841.0	2803.1	2.5547	3.7028	6.2575					
3.0	233.90	0.001 217	0.066 68	1004.78	1599.3	2604.1	1008.42	1795.7	2804.2	2.6457	3.5412	6.1869					
3.5	242.60	0.001 235	0.057 07	1045.43	1558.3	2603.7	1049.75	1753.7	2803.4	2.7253	3.4000	6.1253					
4	250.40	0.001 252	0.049 78	1082.31	1520.0	2602.3	1087.31	1714.1	2801.4	2.7964	3.2737	6.0701					
5	263.99	0.001 286	0.039 44	1147.81	1449.3	2597.1	1154.23	1640.1	2794.3	2.9202	3.0532	5.9734					
6	275.64	0.001 319	0.032 44	1205.44	1384.3	2589.7	1213.35	1571.0	2784.3	3.0267	2.8625	5.8892					
7	285.88	0.001 351	0.027 37	1257.55	1323.0	2580.5	1267.00	1505.1	2772.1	3.1211	2.6922	5.8133					
8	295.06	0.001 384	0.023 52	1305.57	1264.2	2569.8	1316.64	1441.3	2758.0	3.2068	2.5364	5.7432					
9	303.40	0.001 418	0.020 48	1350.51	1207.3	2557.8	1363.26	1378.9	2742.1	3.2858	2.3915	5.6772					
10	311.06	0.001 452	0.018 026	1393.04	1151.4	2544.4	1407.56	1317.1	2724.7	3.3596	2.2544	5.6141					
11	318.15	0.001 489	0.015 987	1433.7	1096.0	2529.8	1450.1	1255.5	2705.6	3.4295	2.1233	5.5527					
12	324.75	0.001 527	0.014 263	1473.0	1040.7	2513.7	1491.3	1193.6	2684.9	3.4962	1.9962	5.4924					
13	330.93	0.001 567	0.012 780	1511.1	985.0	2496.1	1531.5	1130.7	2662.2	3.5606	1.8718	5.4323					
14	336.75	0.001 611	0.011 485	1548.6	928.2	2476.8	1571.1	1066.5	2637.6	3.6232	1.7485	5.3717					
15	342.24	0.001 658	0.010 337	1585.6	869.8	2455.5	1610.5	1000.0	2610.5	3.6848	1.6249	5.3098					
16	347.44	0.001 711	0.009 306	1622.7	809.0	2431.7	1650.1	930.6	2580.6	3.7461	1.4994	5.2455					
17	352.37	0.001 770	0.008 364	1660.2	744.8	2405.0	1690.3	856.9	2547.2	3.8079	1.3698	5.1777					
18	357.06	0.001 840	0.007 489	1698.9	675.4	2374.3	1732.0	777.1	2509.1	3.8715	1.2329	5.1044					
19	361.54	0.001 924	0.006 657	1739.9	598.1	2338.1	1776.5	688.0	2464.5	3.9388	1.0839	5.0228					
20	365.81	0.002 036	0.005 834	1785.6	507.5	2293.0	1826.3	583.4	2409.7	4.0139	0.9130	4.9269					
21	369.89	0.002 207	0.004 952	1842.1	388.5	2230.6	1888.4	446.2	2334.6	4.1075	0.6938	4.8013					
22	373.80	0.002 742	0.003 568	1961.9	125.2	2087.1	2022.2	143.4	2165.6	4.3110	0.2216	4.5327					
22.09	374.14	0.003 155	0.003 155	2029.6	0	2029.6	2099.3	0	2099.3	4.4298	0	4.4298					

TABLE A.3 (SI)
Properties of Superheated Steam

T	P = .010 MPa (45.81)					P = .050 MPa (81.33)					P = .10 MPa (99.63)					
	v	u	h	s	v	u	h	s	v	u	h	s	v	u	h	s
Sat.	14.674	2437.9	2584.7	8.1502	3.240	2483.9	2645.9	7.5939	1.6940	2506.1	2675.5	7.3594				
50	14.869	2443.9	2592.6	8.1749	3.418	2511.6	2682.5	7.6947	1.6958	2506.7	2676.2	7.3614				
100	17.196	2515.5	2687.5	8.4479	3.889	2585.6	2780.1	7.9401	1.9364	2582.8	2776.4	7.6134				
150	19.512	2587.9	2783.0	8.6882	4.356	2659.9	2877.7	8.1580	2.172	2658.1	2875.3	7.8343				
200	21.825	2661.3	2879.5	8.9038	4.820	2735.0	2976.0	8.3556	2.406	2733.7	2974.3	8.0333				
250	24.136	2736.0	2977.3	9.1002	5.284	2811.3	3075.5	8.5373	2.639	2810.4	3074.3	8.2158				
300	26.445	2812.1	3076.5	9.2813	6.209	2968.5	3278.9	8.8642	3.103	2967.9	3278.2	8.5435				
400	31.063	2968.9	3279.6	9.6077	7.134	3132.0	3488.7	9.1546	3.565	3131.6	3488.1	8.8342				
500	35.679	3132.3	3489.1	9.8978	8.057	3302.2	3705.1	9.4178	4.028	3301.9	3704.7	9.0976				
600	40.295	3302.5	3705.4	10.1608	8.981	3479.4	3928.5	9.6599	4.490	3479.2	3928.2	9.3398				
700	44.911	3479.6	3928.7	10.4028	9.904	3663.6	4158.9	9.8852	4.952	3663.5	4158.6	9.5652				
800	49.526	3663.8	4159.0	10.6281	10.828	3854.9	4396.3	10.0967	5.414	3854.8	4396.1	9.7767				
900	54.141	3855.0	4396.4	10.8396	11.751	4052.9	4640.5	10.2964	5.875	4052.8	4640.3	9.9764				
1000	58.757	4053.0	4640.6	11.0393	12.674	4257.4	4891.1	10.4859	6.337	4257.3	4891.0	10.1659				
1100	63.372	4257.5	4891.2	11.2287	13.597	4467.8	5147.7	10.6662	6.799	4467.7	5147.6	10.3463				
1200	67.987	4467.9	5147.8	11.4091	14.521	4683.6	5409.6	10.8382	7.260	4683.5	5409.5	10.5183				
1300	72.602	4683.7	5409.7	11.5811												
					P = .30 MPa (133.55)					P = .40 MPa (143.63)						
					v	u	h	s	v	u	h	s	v	u	h	s
Sat.	.8857	2529.5	2706.7	7.1272	.6058	2543.6	2725.3	6.9919	.4625	2553.6	2738.6	6.8959				
150	.9596	2576.9	2768.8	7.2795	.6339	2570.8	2761.0	7.0778	.4708	2564.5	2752.8	6.9299				
200	1.0803	2654.4	2870.5	7.5066	.7163	2650.7	2865.6	7.3115	.5342	2646.8	2860.5	7.1706				
250	1.1988	2731.2	2971.0	7.7086	.7964	2728.7	2967.6	7.5166	.5951	2726.1	2964.2	7.3789				
300	1.3162	2808.6	3071.8	7.8926	.8753	2806.7	3069.3	7.7022	.6548	2804.8	3066.8	7.5662				
400	1.5493	2966.7	3276.6	8.2218	1.0315	2965.6	3275.0	8.0330	.7726	2964.4	3273.4	7.8985				

TABLE A.3 (SI) (cont'd.)

T	P = 25.0 MPa					P = 30.0 MPa					P = 35.0 MPa					
	v	u	h	s	v	u	h	s	v	u	h	s	v	u	h	s
700	.016 646	3361.3	3777.5	6.6707	.013 661	3335.8	3745.6	6.5606	.011 533	3309.8	3713.5	6.4631	.011 533	3309.8	3713.5	6.4631
800	.018 912	3574.3	4047.1	6.9345	.015 623	3555.5	4024.2	6.8332	.013 278	3536.7	4001.5	6.7450	.013 278	3536.7	4001.5	6.7450
900	.021 045	3783.0	4309.1	7.1680	.017 448	3768.5	4291.9	7.0718	.014 883	3754.0	4274.9	6.9886	.014 883	3754.0	4274.9	6.9886
1000	.023 10	3990.9	4568.5	7.3802	.019 196	3978.8	4554.7	7.2867	.016 410	3966.7	4541.1	7.2064	.016 410	3966.7	4541.1	7.2064
1100	.025 12	4200.2	4828.2	7.5765	.020 903	4189.2	4816.3	7.4845	.017 895	4178.3	4804.6	7.4057	.017 895	4178.3	4804.6	7.4057
1200	.027 11	4412.0	5089.9	7.7605	.022 589	4401.3	5079.0	7.6692	.019 360	4390.7	5068.3	7.5910	.019 360	4390.7	5068.3	7.5910
1300	.029 10	4626.9	5354.4	7.9342	.024 266	4616.0	5344.0	7.8432	.020 815	4605.1	5333.6	7.7653	.020 815	4605.1	5333.6	7.7653
P = 40.0 MPa																
375	.001 640 7	1677.1	1742.8	3.8290	.001 559 4	1638.6	1716.6	3.7639	.001 502 8	1609.4	1699.5	3.7141	.001 502 8	1609.4	1699.5	3.7141
400	.001 907 7	1854.6	1930.9	4.1135	.001 730 9	1788.1	1874.6	4.0031	.001 633 5	1745.4	1843.4	3.9318	.001 633 5	1745.4	1843.4	3.9318
425	.002 532	2096.9	2198.1	4.5029	.002 007	1959.7	2060.0	4.2734	.001 816 5	1892.7	2001.7	4.1626	.001 816 5	1892.7	2001.7	4.1626
450	.003 693	2365.1	2512.8	4.9459	.002 486	2159.6	2284.0	4.5884	.002 085	2053.9	2179.0	4.4121	.002 085	2053.9	2179.0	4.4121
500	.005 622	2678.4	2903.3	5.4700	.003 892	2525.5	2720.1	5.1726	.002 956	2390.6	2567.9	4.9321	.002 956	2390.6	2567.9	4.9321
550	.006 984	2869.7	3149.1	5.7785	.005 118	2763.6	3019.5	5.5485	.003 956	2658.8	2896.2	5.3441	.003 956	2658.8	2896.2	5.3441
600	.008 094	3022.6	3346.4	6.0114	.006 112	2942.0	3247.6	5.8178	.004 834	2861.1	3151.2	5.6452	.004 834	2861.1	3151.2	5.6452
650	.009 063	3158.0	3520.6	6.2054	.006 966	3093.5	3441.8	6.0342	.005 595	3028.8	3364.5	5.8829	.005 595	3028.8	3364.5	5.8829
700	.009 941	3283.6	3681.2	6.3750	.007 727	3230.5	3616.8	6.2189	.006 272	3177.2	3553.5	6.0824	.006 272	3177.2	3553.5	6.0824
800	.011 523	3517.8	3978.7	6.6662	.009 076	3479.8	3933.6	6.5290	.007 459	3441.5	3889.1	6.4109	.007 459	3441.5	3889.1	6.4109
900	.012 962	3739.4	4257.9	6.9150	.010 283	3710.3	4224.4	6.7882	.008 508	3681.0	4191.5	6.6805	.008 508	3681.0	4191.5	6.6805
1000	.014 324	3954.6	4527.6	7.1356	.011 411	3930.5	4501.1	7.0146	.009 480	3906.4	4475.2	6.9127	.009 480	3906.4	4475.2	6.9127
1100	.015 642	4167.4	4793.1	7.3364	.012 496	4145.7	4770.5	7.2184	.010 409	4124.1	4748.6	7.1195	.010 409	4124.1	4748.6	7.1195
1200	.016 940	4380.1	5057.7	7.5224	.013 561	4359.1	5037.2	7.4058	.011 317	4338.2	5017.2	7.3083	.011 317	4338.2	5017.2	7.3083
1300	.018 229	4594.3	5323.5	7.6969	.014 616	4572.8	5303.6	7.5808	.012 215	4551.4	5284.3	7.4837	.012 215	4551.4	5284.3	7.4837

TABLE 4

<i>p</i> (t Sat.) MPa	Liquid											
	0			2.5 (223.99)			5.0 (263.99)					
<i>t</i>	$10^3 v$	μ	<i>h</i>	<i>s</i>	$10^3 v$	μ	<i>h</i>	<i>s</i>	$10^3 v$	μ	<i>h</i>	<i>s</i>
Sat.												
0	1.0002	-0.03	-0.03	-0.0001	0.9990	-0.00	2.50	-0.0000	0.9977	0.04	5.04	0.0001
20	1.0018	83.95	83.95	0.2966	1.0006	83.80	86.30	0.2961	0.9995	83.65	88.65	0.2956
40	1.0078	167.56	167.56	0.5725	1.0067	167.25	169.77	0.5715	1.0056	166.95	171.97	0.5705
60	1.0172	251.12	251.12	0.8312	1.0160	250.67	253.21	0.8298	1.0149	250.23	255.30	0.8285
80	1.1291	334.87	334.87	1.0753	1.0280	334.29	336.86	1.0737	1.0268	333.72	338.85	1.0720
100	1.0436	418.96	418.96	1.3069	1.0423	418.24	420.85	1.3050	1.0410	417.52	422.72	1.3030
120	1.0604	503.57	503.57	1.5278	1.0590	502.68	505.33	1.5255	1.0576	501.80	507.09	1.5233
140	1.0800	588.89	588.89	1.7395	1.0784	587.82	590.52	1.7369	1.0768	586.76	592.15	1.7343
160	1.1024	675.19	675.19	1.9434	1.1006	673.90	676.65	1.9404	1.0988	672.62	678.12	1.9375
180	1.1283	762.72	762.72	2.1410	1.1261	761.16	763.97	2.1375	1.1240	759.63	765.25	2.1341
200	1.1581	851.8	851.8	2.3334	1.1555	849.9	852.8	2.3294	1.1530	848.1	853.9	2.3255
210	1.1749	897.1	897.1	2.4281	1.1720	895.0	898.0	2.4238	1.1691	893.0	898.8	2.4195
220	1.1930	943.0	943.0	2.5221	1.1898	940.7	943.7	2.5174	1.1866	938.4	944.4	2.5128
230	1.2129	989.6	989.6	2.6157	1.2092	987.0	990.1	2.6105	1.2056	984.5	990.6	2.6055
240	1.2347	1037.1	1037.1	2.7091	1.2305	1034.2	1037.2	2.7034	1.2264	1031.4	1037.5	2.6979
250	1.2590	1085.6	1085.6	2.8027	1.2540	1082.3	1085.4	2.7964	1.2493	1079.1	1085.3	2.7902
260	1.2862	1135.4	1135.4	2.8970	1.2804	1131.6	1134.8	2.8898	1.2749	1127.9	1134.3	2.8830
270	1.3173	1186.8	1186.8	2.9926	1.3102	1182.4	1185.7	2.9844	1.3036	1178.2	1184.3	2.9766
280	1.3535	1240.4	1240.4	3.0904	1.3447	1235.1	1238.5	3.0808	1.3365	1230.2	1236.8	3.0717
290	1.3971	1297.0	1297.0	3.1918	1.3855	1290.5	1294.0	3.1801	1.3750	1284.4	1291.3	3.1693
300	1.4520	1358.1	1358.1	3.2992	1.4357	1349.6	1353.2	3.2843	1.4214	1341.9	1349.0	3.2708
310									1.4803	1404.1	1411.5	3.3789

FIGURE 5.11a Extract from subcooled table (SI units).

TABLE A.4 (SI)
Properties of Compressed Liquid (Steam)

T	P = 5 MPa (263.99)					P = 10 MPa (311.06)					P = 15 MPa (342.24)				
	v	u	h	s		v	u	h	s		v	u	h	s	
Sat.	.001 285 9	1147.8	1154.2	2.9202		.001 452 4	1393.0	1407.6	3.3596		.001 658 1	1585.6	1610.5	3.6848	
0	.000 997 7	.04	5.04	.0001		.000 995 2	.09	10.04	.0002		.000 992 8	.15	15.05	.0004	
20	.000 999 5	83.65	88.65	.2956		.000 997 2	83.36	93.33	.2945		.000 995 0	83.06	97.99	.2934	
40	.001 005 6	166.95	171.97	.5705		.001 003 4	166.35	176.38	.5686		.001 001 3	165.76	180.78	.5666	
60	.001 014 9	250.23	255.30	.8285		.001 012 7	249.36	259.49	.8258		.001 010 5	248.51	263.67	.8232	
80	.001 026 8	333.72	338.85	1.0720		.001 024 5	332.59	342.83	1.0688		.001 022 2	331.48	346.81	1.0656	
100	.001 041 0	417.52	422.72	1.3030		.001 038 5	416.12	426.50	1.2992		.001 036 1	414.74	430.28	1.2955	
120	.001 057 6	501.80	507.09	1.5233		.001 054 9	500.08	510.64	1.5189		.001 052 2	498.40	514.19	1.5145	
140	.001 076 8	586.76	592.15	1.7343		.001 073 7	584.68	595.42	1.7292		.001 070 7	582.66	598.72	1.7242	
160	.001 098 8	672.62	678.12	1.9375		.001 095 3	670.13	681.08	1.9317		.001 091 8	667.71	684.09	1.9260	
180	.001 124 0	759.63	765.25	2.1341		.001 119 9	756.65	767.84	2.1275		.001 115 9	753.76	770.50	2.1210	
200	.001 153 0	848.1	853.9	2.3255		.001 148 0	844.5	856.0	2.3178		.001 143 3	841.0	858.2	2.3104	
220	.001 186 6	938.4	944.4	2.5128		.001 180 5	934.1	945.9	2.5039		.001 174 8	929.9	947.5	2.4953	
240	.001 226 4	1031.4	1037.5	2.6979		.001 218 7	1026.0	1038.1	2.6872		.001 211 4	1020.8	1039.0	2.6771	
260	.001 274 9	1127.9	1134.3	2.8830		.001 264 5	1121.1	1133.7	2.8699		.001 255 0	1114.6	1133.4	2.8576	
280						.001 321 6	1220.9	1234.1	3.0548		.001 308 4	1212.5	1232.1	3.0393	
300						.001 397 2	1328.4	1342.3	3.2469		.001 377 0	1316.6	1337.3	3.2260	
320											.001 472 4	1431.1	1453.2	3.4247	
340											.001 631 1	1567.5	1591.9	3.6546	

TABLE A.4 (SI) (cont'd.)

T	P = 20 MPa (365.81)					P = 30 MPa					P = 50 MPa				
	v	u	h	s	s	v	u	h	s	s	v	u	h	s	s
Sat.	.002 036	1785.6	1826.3	4.0139		.000 985 6	.25	29.82	.0001		.000 976 6	.20	49.03	.0014	
0	.000 990 4	.19	20.01	.0004		.000 988 6	82.17	111.84	.2899		.000 980 4	81.00	130.02	.2848	
20	.000 992 8	82.77	102.62	.2923		.000 995 1	164.04	193.89	.5607		.000 987 2	161.86	211.21	.5527	
40	.000 999 2	165.17	185.16	.5646		.001 004 2	246.06	276.19	.8154		.000 996 2	242.98	292.79	.8052	
60	.001 008 4	247.68	267.85	.8206		.001 015 6	328.30	358.77	1.0561		.001 007 3	324.34	374.70	1.0440	
80	.001 019 9	330.40	350.80	1.0624		.001 029 0	410.78	441.66	1.2844		.001 020 1	405.88	456.89	1.2703	
100	.001 033 7	413.39	434.06	1.2917		.001 044 5	493.59	524.93	1.5018		.001 034 8	487.65	539.39	1.4857	
120	.001 049 6	496.76	517.76	1.5102		.001 062 1	576.88	608.75	1.7098		.001 051 5	569.77	622.35	1.6915	
140	.001 067 8	580.69	602.04	1.7193		.001 082 1	660.82	693.28	1.9096		.001 070 3	652.41	705.92	1.8891	
160	.001 088 5	665.35	687.12	1.9204		.001 104 7	745.59	778.73	2.1024		.001 091 2	735.69	790.25	2.0794	
180	.001 112 0	750.95	773.20	2.1147		.001 130 2	831.4	865.3	2.2893		.001 114 6	819.7	875.5	2.2634	
200	.001 138 8	837.7	860.5	2.3031		.001 159 0	918.3	953.1	2.4711		.001 140 8	904.7	961.7	2.4419	
220	.001 169 3	925.9	949.3	2.4870		.001 192 0	1006.9	1042.6	2.6490		.001 170 2	990.7	1049.2	2.6158	
240	.001 204 6	1016.0	1040.0	2.6674		.001 230 3	1097.4	1134.3	2.8243		.001 203 4	1078.1	1138.2	2.7860	
260	.001 246 2	1108.6	1133.5	2.8459		.001 275 5	1190.7	1229.0	2.9986		.001 241 5	1167.2	1229.3	2.9537	
280	.001 296 5	1204.7	1230.6	3.0248		.001 330 4	1287.9	1327.8	3.1741		.001 286 0	1258.7	1323.0	3.1200	
300	.001 359 6	1306.1	1333.3	3.2071		.001 399 7	1390.7	1432.7	3.3539		.001 338 8	1353.3	1420.2	3.2868	
320	.001 443 7	1415.7	1444.6	3.3979		.001 492 0	1501.7	1546.5	3.5426		.001 403 2	1452.0	1522.1	3.4557	
340	.001 568 4	1539.7	1571.0	3.6075		.001 626 5	1626.6	1675.4	3.7494		.001 483 8	1556.0	1630.2	3.6291	
360	.001 822 6	1702.8	1739.3	3.8772		.001 869 1	1781.4	1837.5	4.0012		.001 588 4	1667.2	1746.6	3.8101	
380															