

Professional Engineers Ontario

National Examinations December 2019

16-Civ-B9, The Finite Element Method

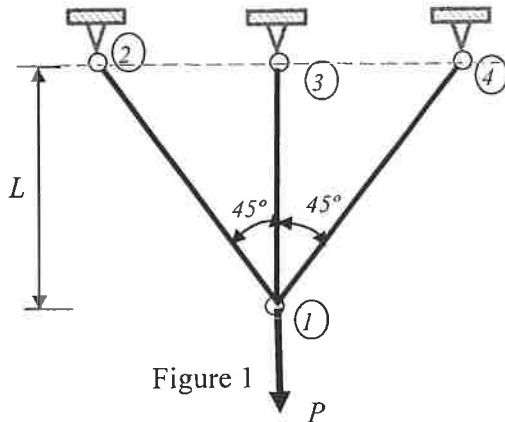
3 hours duration

Notes:

1. There are 4 pages in this examination.
 2. If doubt exists as to the interpretation of any question, the candidate is urged to submit with the answer paper, a clear statement of any assumptions made
 3. This is a closed book exam but one aid sheet allowed written on both sides.
 4. Candidates may use a Casio or Sharp approved calculator.
 5. **Answer all proposed problems**
 6. Marking scheme is shown on the exam paper.
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Problem 1 (25 Marks)

A three-bar truss is shown in Figure 1. All bars have the same axial stiffness EA . When the temperature was at a reference value T_0 , the truss was subjected to a vertical force P . The temperature of bar 1-2 is increased by an amount ΔT above T_0 . The thermal coefficients of all bars are equal and denoted α .



$$L = 2000 \text{ mm}, A = 3000 \text{ mm}^2$$

$$E = 200 \text{ GPa}$$

$$\Delta T = 30^\circ, \alpha = 12 \times 10^{-6} / ^\circ\text{C}$$

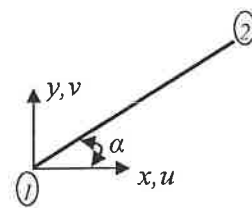
Figure 1

1.1 Determine the displacement of joint 1 and the forces in the three bars when the system reaches equilibrium.

1.2 Using this example as a hint, explain how you proceed to analyse a 3D structure subjected to a mechanical force system as well as a temperature gradient.

The stiffness matrix of a bar with orientation defined by an angle α with respect to the global reference system (x, y) is given by:

$$[k] = \frac{EA}{L} \begin{bmatrix} C^2 & CS & -C^2 & -CS \\ CS & S^2 & -CS & -S^2 \\ -C^2 & -CS & C^2 & CS \\ -CS & -S^2 & CS & S^2 \end{bmatrix} \quad C = \cos(\alpha), \quad S = \sin(\alpha)$$



The internal force in the bar is given by:

$$F_{(1,2)} = \frac{EA}{L} [C \quad S] \begin{bmatrix} u_2 - u_1 \\ v_2 - v_1 \end{bmatrix}$$

where u_1, v_1 are the displacement of node 1 in the global reference and u_2, v_2 are the displacement of node 2 in the global reference

Problem 2 (25 marks)

A beam is connected to two truss bars is shown in Figure 2. The beam is considered inextensible (axial rigidity is infinite) with bending rigidity EI . The truss bars are assumed to have hinge connections at the both ends. The axial rigidity EA of both truss bars is assumed to be equal to $24 \frac{EI}{L^2}$.

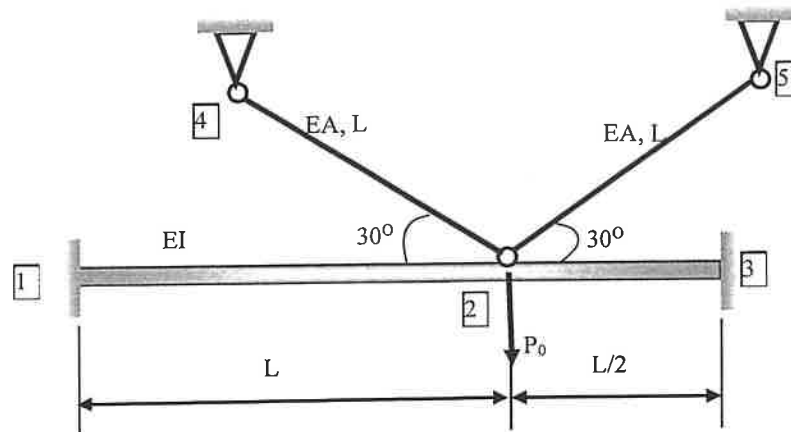


Figure 2

- 2.1 Using a discretization with two beam and two truss elements, define the active DOF of the system.
- 2.2 Assemble the stiffness matrix of the system
- 2.3 Calculate the displacements in the system.
- 2.4 Calculate the axial forces in the two truss bars.

The stiffness matrix of a beam and truss elements are given by:

$$[K] = \frac{EI}{L^3} \begin{bmatrix} 12 & 6L & -12 & 6L \\ & 4L^2 & -6L & 2L^2 \\ SYM & & 12 & -6L \\ & & & 4L^2 \end{bmatrix}$$

Problem 3 (50 marks)

3.1 Referring to Figure 3(a), the interpolation of the displacement field inside the Q4 element is given by:

$$u(x, y) = \sum_{i=1}^4 N_i(x, y)u_i \quad \text{and} \quad v(x, y) = \sum_{i=1}^4 N_i(x, y)v_i$$

Show that the shape function N_i are given by:

$$N_1 = \left(1 - \frac{x}{L}\right)\left(1 - \frac{y}{L}\right); \quad N_2 = \frac{x}{L}\left(1 - \frac{y}{L}\right); \quad N_3 = \frac{xy}{L^2}; \quad N_4 = \left(1 - \frac{x}{L}\right)\frac{y}{L}$$

3.2 Derive the strain-displacement matrix [B] defined from:

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{Bmatrix} = [B]\{d\} \quad \text{and} \quad \text{vector } \{d\} = \{u_1, v_1, u_2, v_2, u_3, v_3, u_4, v_4\}^T$$

3.3 Show that in the case of plane stress, the linear relationship between the horizontal forces $\{F_2, F_3\}$ and displacements of node 2 and 3, $\{u_2, u_3\}$ is given by:

$$\begin{Bmatrix} F_2 \\ F_3 \end{Bmatrix} = \begin{bmatrix} k_{11} & k_{12} \\ k_{12} & k_{22} \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix}$$

where $k_{11} = k_{22} = \frac{Et}{1-\nu^2} \left[\frac{1}{3} + \frac{1-\nu}{6} \right]$, $k_{12} = \frac{Et}{1-\nu^2} \left[\frac{1}{6} + \frac{1-\nu}{6} \right]$. t is the thickness of the element.

The element is made of an elastic material with an elasticity modulus E and Poisson's ratio ν .

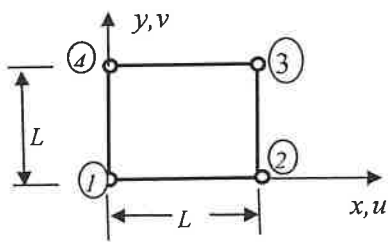


Fig. 3(a)

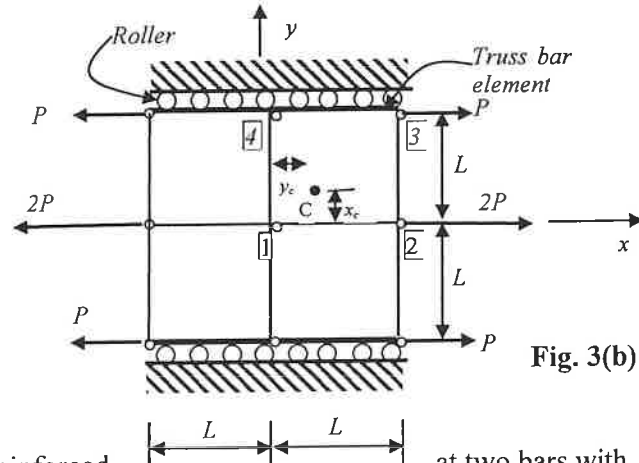


Fig. 3(b)

3.4 Figure 3(b) shows a plate of thickness t reinforced at two bars with constant cross sections A . The plate is modeled using four rectangular elements. Assuming that $Et = \frac{2EA}{L}$.

Calculate displacements at node 2 and 3.

Calculate the strains at point C, $x_c = y_c = \frac{L}{4}$.

Use the following data: $E = 200 \text{ GPa}$, $\nu = 0.25$, $L = 20 \text{ mm}$, $t = 5 \text{ mm}$ and $P = 5 \text{ kN}$.